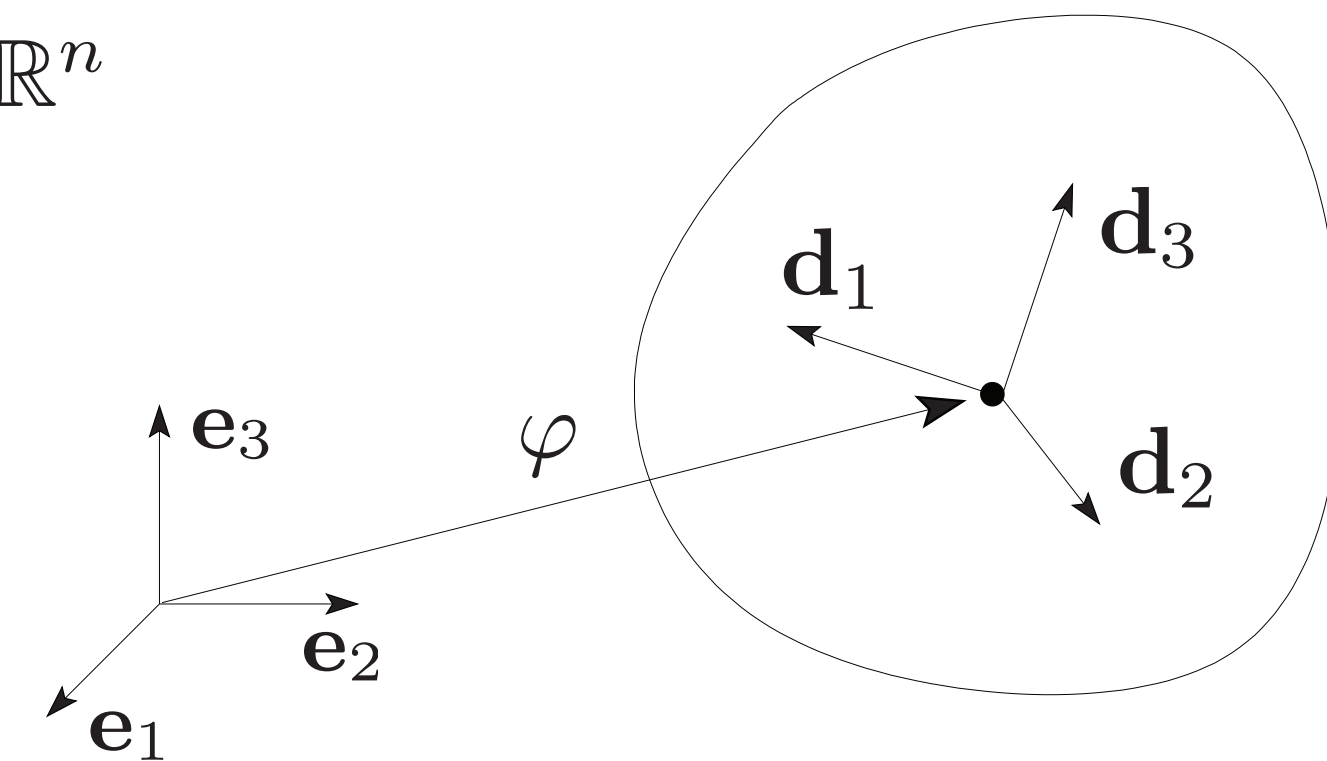


## rigid body kinematics

- configuration  $\mathbf{q} : [0, T] \rightarrow \mathbb{R}^n$

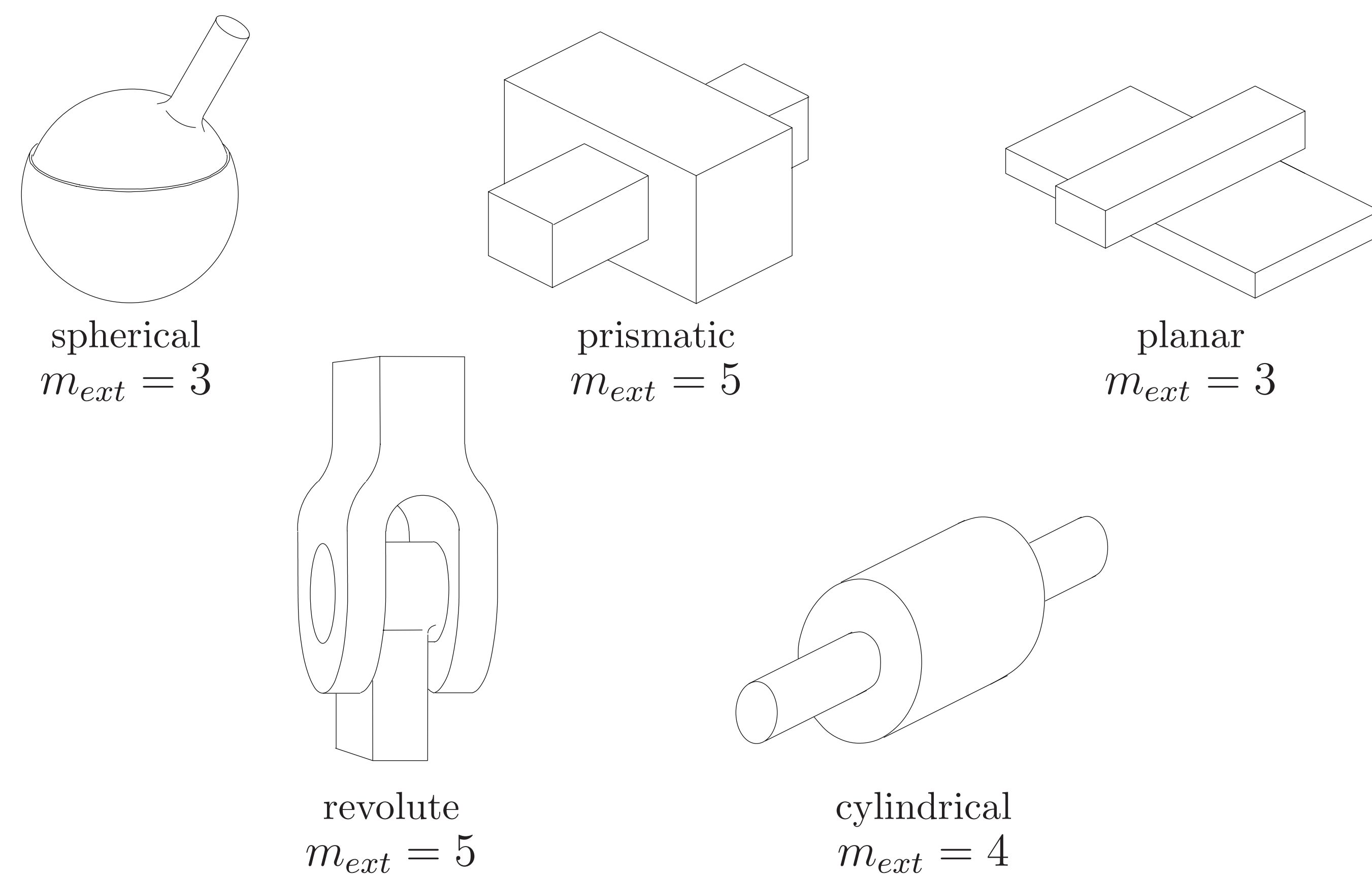
$$\mathbf{q}(t) = [\varphi, \mathbf{d}_1, \mathbf{d}_2, \mathbf{d}_3](t)$$



- internal constraints: **orthonormal** director triad

$$g_i(\mathbf{q}) = \mathbf{d}_j^T \cdot \mathbf{d}_k - \delta_{jk} = 0 \quad j, k = 1, 2, 3 \quad i = 1, \dots, m_{int}$$

- ⊕ directors allow straightforward formulation of external constraints: **joints in multibody systems**



- holonomic constraints  $\mathbf{g}(\mathbf{q}) \in \mathbb{R}^m \quad m = m_{int} + m_{ext}$

## variational discrete null space method

- ⊕ easy formulation of discrete Lagrangian  $L_d(\mathbf{q}_n, \mathbf{q}_{n+1})$  with **constant** mass matrix
- discrete **nodal** reparametrisation  $\mathbf{q}_n = \mathbf{F}(\mathbf{u}_n, \mathbf{q}_{n-1})$  such that constraints are fulfilled

- discrete variational principle leads to

$$\left( \frac{\partial \mathbf{F}}{\partial \mathbf{u}_n} \right)^T \cdot [D_1 L_d(\mathbf{q}_n, \mathbf{q}_{n+1}) + D_2 L_d(\mathbf{q}_{n-1}, \mathbf{q}_n)] = \mathbf{0}$$

to be solved for discrete generalised configuration  $\mathbf{u}_{n+1}$

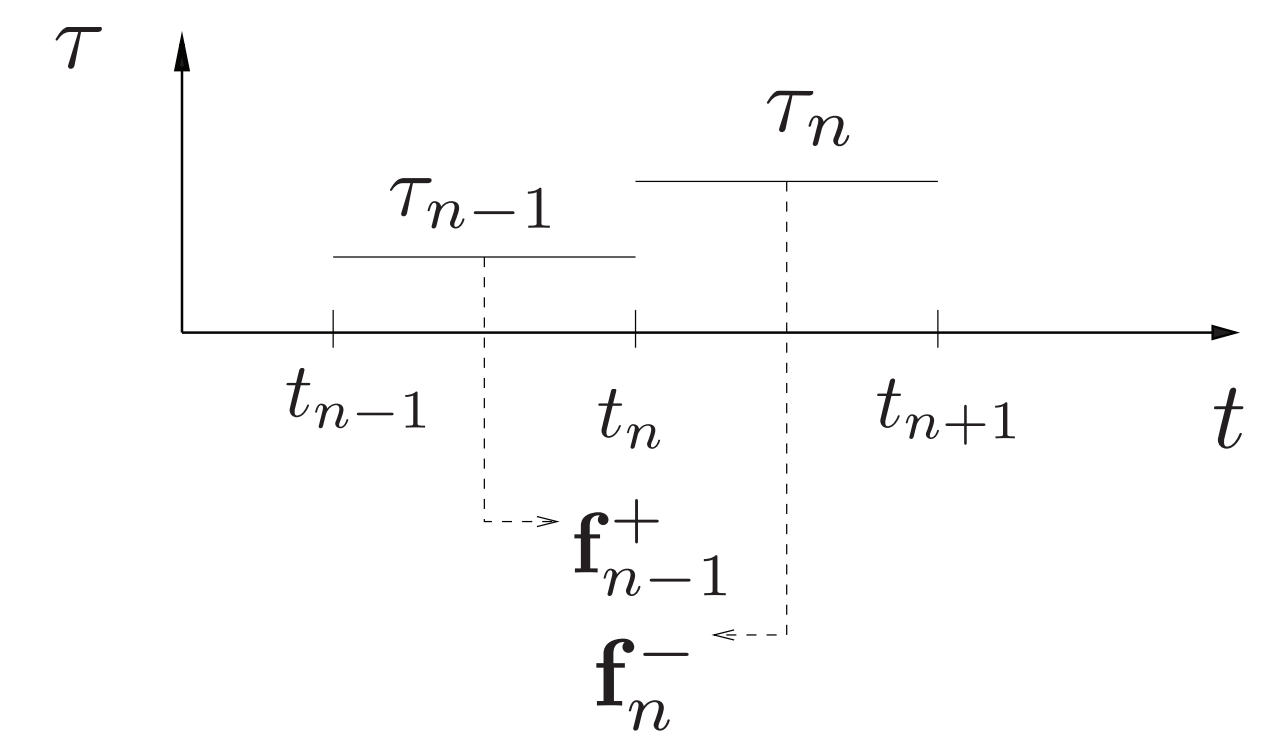
- ⊕ symplectic-momentum conservation
- ⊕ exact constraint fulfilment
- ⊕ **minimal** dimension of system:  $n - m$
- ⊕ conditioning of iteration matrix is independent of  $\Delta t$

## unlike

- Lagrange multiplier method
  - ⊖ **maximal** dimension of system:  $n + m$
  - ⊖ conditioning of iteration matrix is  $\mathcal{O}(1/\Delta t^3)$
- generalised coordinates
  - ⊖ complicated Lagrangian, higher **nonlinearity**
  - ⊖ difficult for multibody systems

## discrete mechanics and optimal control

- discrete control forces  $\mathbf{f}_{n-1}^+(\tau_{n-1}) \quad \mathbf{f}_n^-(\tau_n)$
- discrete generalised force  $\tau$



- ⊕ minimise control effort

$$J_d = \Delta t \sum_{n=0}^{N-1} \|\tau_n\|^2$$

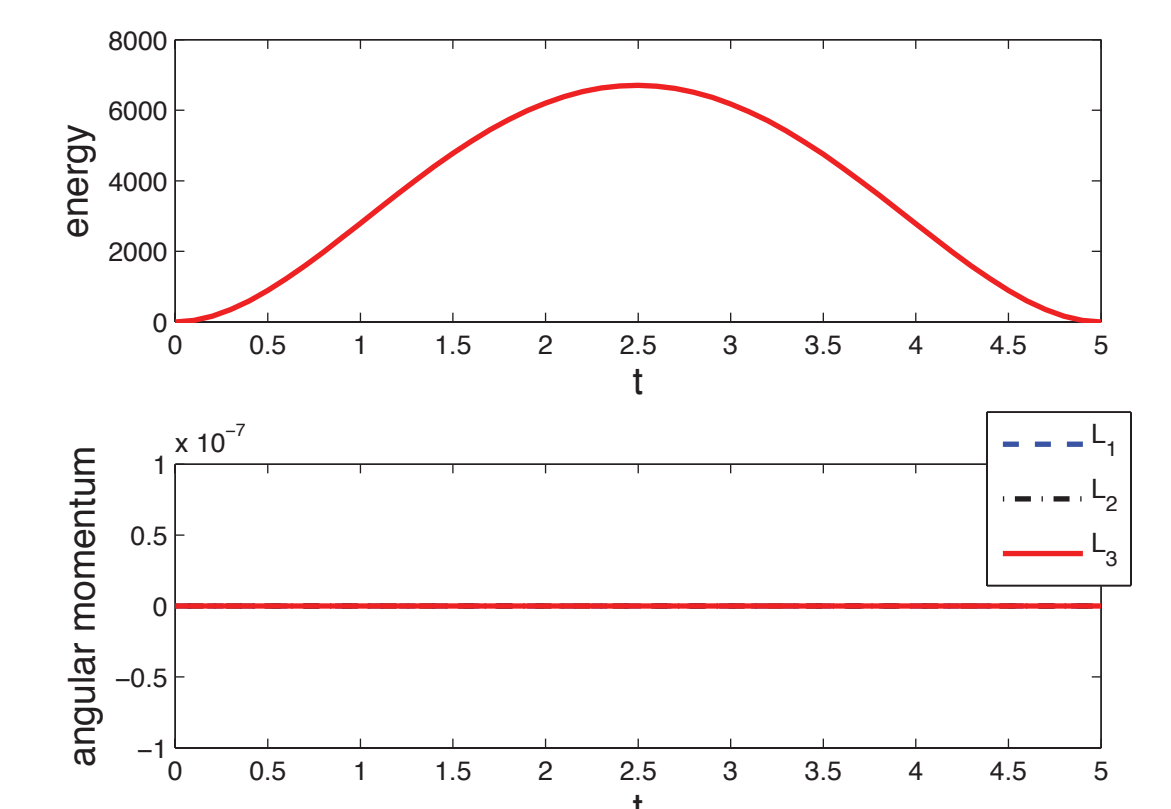
wrt **minimal** possible number of unknowns  $\{\tau_n\}_{n=0}^{N-1}$  subject to

- prescribed initial and final state and
- forced time stepping equations

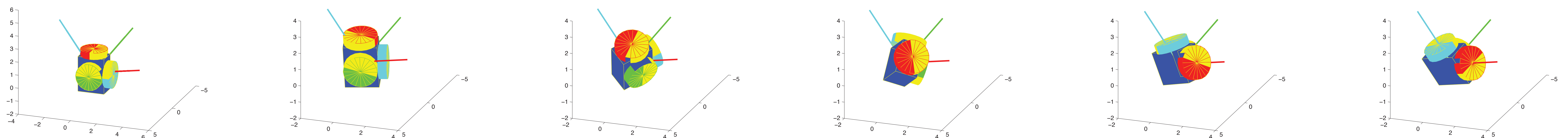
$$\left( \frac{\partial \mathbf{F}}{\partial \mathbf{u}_n} \right)^T \cdot [D_1 L_d(\mathbf{q}_n, \mathbf{q}_{n+1}) + D_2 L_d(\mathbf{q}_{n-1}, \mathbf{q}_n) + \mathbf{f}_{n-1}^+ + \mathbf{f}_n^-] = \mathbf{0}$$

⇒ optimal trajectory is **consistent** with dynamics  
change in angular momentum **exactly** reflects applied torque

- manoeuvre starts and ends at rest



- angular momentum is zero due to **geometric phase**



physically consistent reorientation of 'satellite' with **minimal** control effort